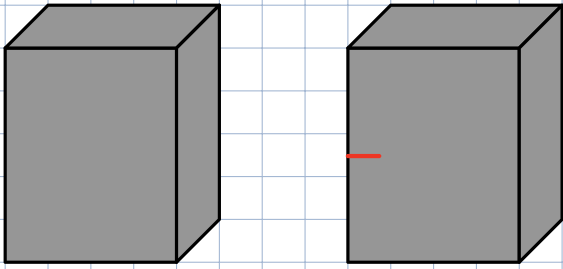


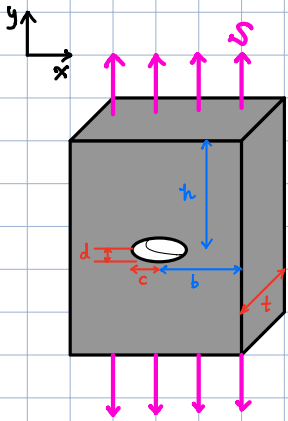
Fracture Mechanics



The presence of a crack \rightarrow fails below σ_y

why? Fracture Mechanics!

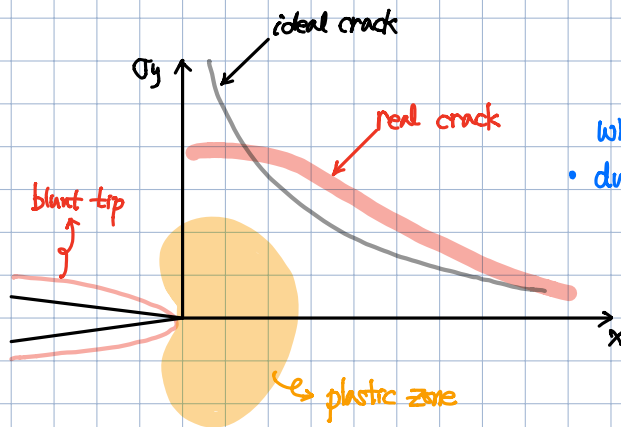
Crack as Stress Raiser



$$\sigma_y = S (1 + 2 \cdot \frac{c}{d}) \leftarrow \text{stress concentration}$$

$$= S (1 + 2 \cdot \sqrt{\frac{c}{\rho}}) \leftarrow \text{tip radius, } \rho = \frac{d^3}{c}$$

$$\therefore \text{stress concentration factor} = k_t = \frac{\sigma_y}{S}$$



why?

• ductile material \rightarrow large plastic deformation

• the crack tip $\rightarrow \sigma \uparrow \rightarrow \sigma > \sigma_y \rightarrow$ plastic zone \rightarrow sharp tip becomes blunt $\rightarrow \sigma = \infty \times$

Stress Analysis of Crack (Stress Intensity Factor, K)

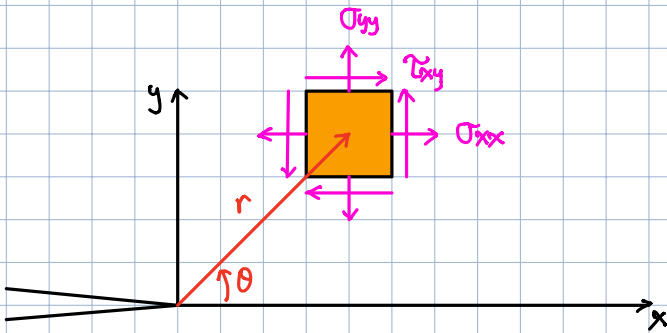
- For certain cracked configuration
- under external forces
- isotropic
- linear elastic
- a polar coordinate axes (origin is at the crack tip)

$$\sigma_{ij} = \left(\frac{K}{\sqrt{r}}\right) f_{ij}(\theta) + \sum_{n=2}^{\infty} A_n \cdot r^{n/2} \cdot g_{ij}^{(n)}(\theta)$$

← Westergaard, Sneddon, Williams

K : a constant
 f_{ij} : a dimensionless function of θ

Each mode of loading produces the $\frac{1}{\sqrt{r}}$ singularity at the crack tip



K & $f_{ij} \rightarrow$ depend on the mode
 $\hookrightarrow K_I, K_{II}, K_{III}$
 $K = k \sqrt{2\pi}$
 (the stress intensity factor)

Mode I

$$\sigma_{xx} = \frac{K_I}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \cdot (1 - \sin \frac{\theta}{2} \cdot \sin 3\frac{\theta}{2}) + \dots$$

$$\sigma_{zz} = 0 \quad (\text{plane stress})$$

$$\sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \cdot (1 + \sin \frac{\theta}{2} \cdot \sin 3\frac{\theta}{2}) + \dots$$

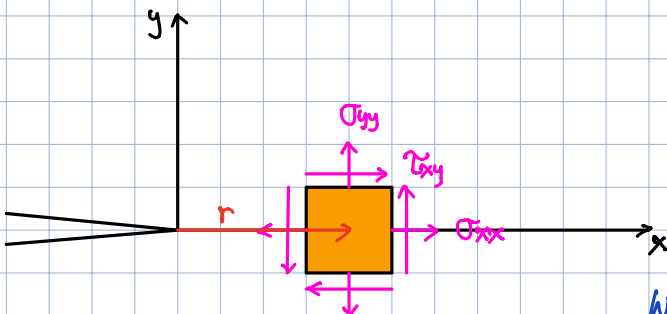
$$\sigma_{zz} = \nu (\sigma_{xx} + \sigma_{yy}) \quad (\text{plane strain: } \epsilon_{zz} = 0)$$

$$\tau_{xy} = \frac{K_I}{\sqrt{2\pi r}} \cos \frac{\theta}{2} \cdot (\sin \frac{\theta}{2} \cdot \sin 3\frac{\theta}{2}) + \dots$$

$$\tau_{yz} = \tau_{zx} = 0$$

(on the crack plane)
 $\theta = 0$;

$$\sigma_{xx} = \sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}}$$



Westergaard solution
 Anderson's textbook
 Appendix A.3.2

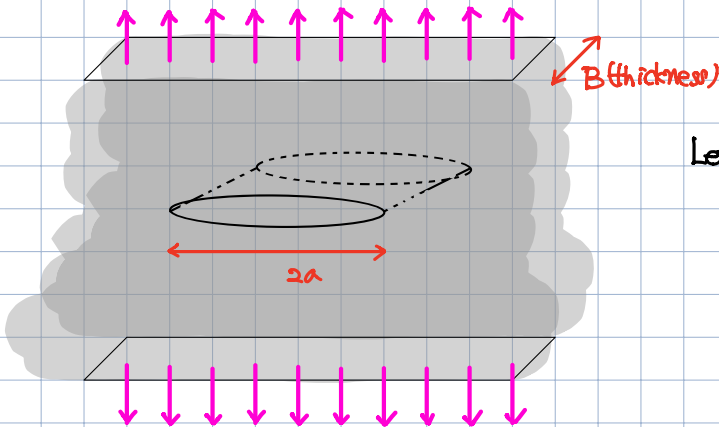
$$K_I = \lim_{r, \theta \rightarrow 0} \sigma_{yy} \sqrt{2\pi r} = FS \sqrt{\pi a}$$

Stress intensity has units of stress $\sqrt{\text{length}}$

Griffith Stress

Assumption: The elastic energy in the elliptical area is released, when the crack with length $2a$ is introduced.

infinite plate



Let's consider a crack of length $2a$ in an "infinite" plate.

Total Energy = (E) Potential Energy + (W_s) Work to create NEW SURFACE

$$E = \pi + W_s$$

$$\frac{dE}{dA} = \frac{d\pi}{dA} + \frac{dW_s}{dA}$$

$$\frac{d\pi}{dA} = -\frac{dW_s}{dA}$$

$$\begin{aligned} \pi &= U - F \\ &= \frac{P\Delta}{2} - P\Delta \\ &= -\frac{P\Delta}{2} = -U \end{aligned}$$

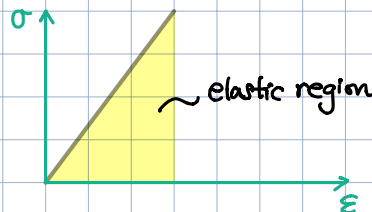
$\gamma_s \sim 1 \text{ J/m}^2$, surface energy
 $\gamma_s \cdot 2aB \times 2 = 4aB\gamma_s$

F : the work done by external forces
 U_E : the strain energy

$$U_E = \frac{\sigma^2}{2E} \cdot \text{Volume} = \frac{\sigma^2}{2E} \cdot (\pi a^2 \cdot B)$$

$$u = \frac{1}{2} \sigma \epsilon = \frac{1}{2} \sigma \cdot (\sigma/E) = \frac{\sigma^2}{2E} \cdot (\pi a^2 \cdot B)$$

$$\therefore U_E = \frac{\sigma^2}{2E} \cdot \text{Volume}$$



$A = 2aB$	$U_E = \frac{\sigma^2}{2E} \cdot (\pi a^2 B)$	$W_s = 4aB \cdot \gamma_s$
$dA/da = 2B$	$dU_E/da = \frac{\sigma^2}{2E} \cdot (2\pi a B)$	$dW_s/da = 4B \cdot \gamma_s$
$\therefore \frac{dU_E}{dA} = \frac{dU_E/da}{dA/da} = \frac{\sigma^2}{E} (\pi a)$, $\frac{dW_s}{da} = 2\gamma_s$		

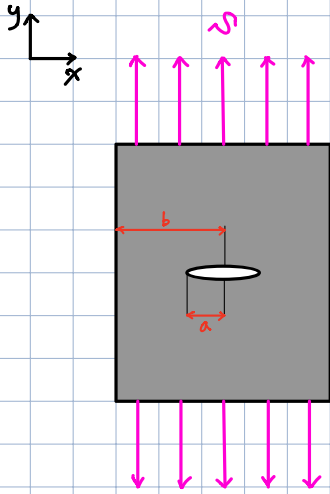
$$\frac{dE}{dA} = \frac{d\pi}{dA} + \frac{dW_s}{dA} = -\frac{\sigma^2}{E} (\pi a) + 2\gamma_s = 0 \quad (\because \text{Energy Balance}) \longrightarrow \sigma^2 = \frac{2E\gamma_s}{\pi a}$$

$$\therefore \sigma_f = \sqrt{\frac{2E\gamma_s}{\pi a}}$$

Ans.

The fracture toughness, K_{Ic}

1. K : the stress intensity factor (LEFM)
2. K_{Ic} : the fracture toughness (vary widely)
3. K_{Ic} : the plane strain fracture toughness (material's ability)



$$K = S \sqrt{\pi a} \quad (a \ll b)$$

For a given material and thickness with K_{Ic}

$$S_c = \frac{K_{Ic}}{\sqrt{\pi a}}$$

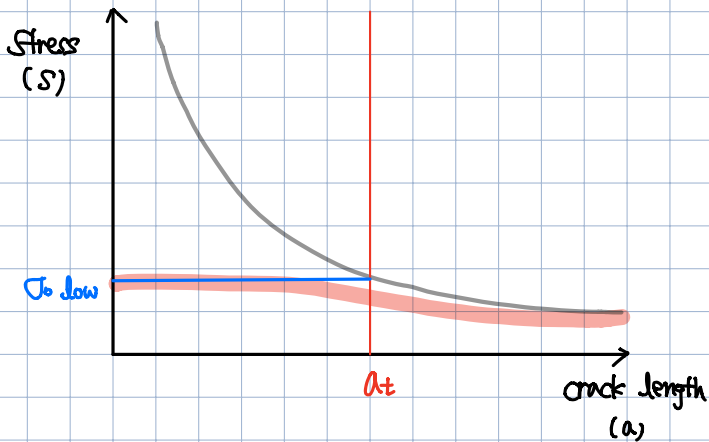
$$\pi a_t = \left(\frac{K_{Ic}}{\sigma_0} \right)^2$$

$$\therefore a_t = \frac{1}{\pi} \cdot \left(\frac{K_{Ic}}{\sigma_0} \right)^2$$

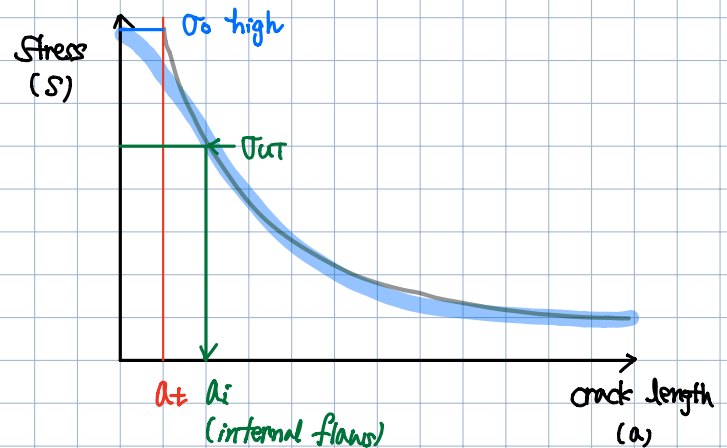
a_t = the transition crack length

my crack size $> a_t$

→ brittle fracture rather yielding.



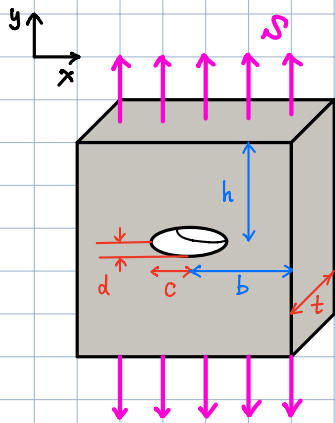
polymer



ceramics

Example >

A center-cracked plate



$b = 50 \text{ mm}$
 $t = 5 \text{ mm}$
 large h
 $P = 50 \text{ kN}$

tip radius, $\rho = d^2/c$

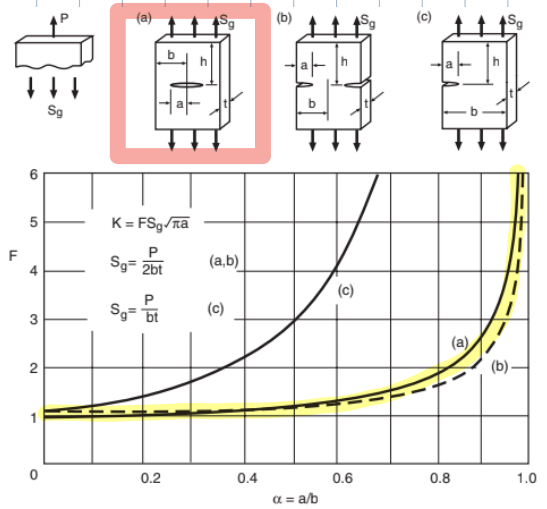
a) $K = ?$ ← When $a = 10 \text{ mm}$
 (the stress intensity factor)

b) $a = 30 \text{ mm}$

c) a_c (for 2014-T61 Aluminum)

$K_{IC} = 24 \text{ MPa}\sqrt{\text{m}}$

F is a dimensionless function that depends on the geometry and loading configuration



Values for small a/b and limits for 10% accuracy:

(a) $K = S_g \sqrt{\pi a}$ (b) $K = 1.12 S_g \sqrt{\pi a}$ (c) $K = 1.12 S_g \sqrt{\pi a}$
 $(a/b \leq 0.4)$ $(a/b \leq 0.6)$ $(a/b \leq 0.13)$

Expressions for any $\alpha = a/b$:

(a) $F = \frac{1 - 0.5\alpha + 0.326\alpha^2}{\sqrt{1 - \alpha}}$ ($h/b \geq 1.5$)
 (b) $F = \left(1 + 0.122 \cos^4 \frac{\pi\alpha}{2}\right) \sqrt{\frac{2}{\pi\alpha} \tan \frac{\pi\alpha}{2}}$ ($h/b \geq 2$)
 (c) $F = 0.265(1 - \alpha)^4 + \frac{0.857 + 0.265\alpha}{(1 - \alpha)^{3/2}}$ ($h/b \geq 1$)

a) $K = F S_g \sqrt{\pi a}$ ← $S_g = \frac{P}{2bt} = \frac{50 \text{ kN}}{(2)(50 \text{ mm})(5 \text{ mm})} = 100 \text{ MPa}$

$\alpha = a/b = \frac{10 \text{ mm}}{50 \text{ mm}} = 0.2 < 0.4 \rightarrow F = 1$

$K = (1) \cdot (100 \text{ MPa}) \cdot \sqrt{(\pi) \cdot (10 \text{ mm})} = 17.7 \text{ MPa}\sqrt{\text{m}}$ Ans.

b) $K = F S_g \sqrt{\pi a}$ ← $S_g = 100 \text{ MPa}$

$\alpha = a/b = \frac{30 \text{ mm}}{50 \text{ mm}} = 0.6 > 0.4 \rightarrow F = \frac{1 - 0.5\alpha + 0.326\alpha^2}{\sqrt{1 - \alpha}}$

$\therefore K = (1.292) \cdot (100 \text{ MPa}) \cdot \sqrt{(\pi) \cdot (30 \text{ mm})} = 39.7 \text{ MPa}\sqrt{\text{m}}$ Ans. $= 1.292$

c) Table \rightarrow 2014-T651 Aluminum $\cdot K_{Ic} = 24 \text{ MPa}\sqrt{\text{m}}$

We don't know $a \rightarrow$ assume that $\alpha = \sigma_b < 0.4 \rightarrow F = 1 \rightarrow K = S_y \sqrt{\pi a}$

$$K/S_y = \sqrt{\pi a}$$

$$a = \frac{1}{\pi} \cdot (K/S_y)^2 = \frac{1}{\pi} \cdot \left(\frac{24 \text{ MPa}\sqrt{\text{m}}}{100 \text{ MPa}} \right)^2 = 0.0183 \text{ m} = \underline{18.3 \text{ mm}} \text{ Ans.}$$

\swarrow with the 10% approximation

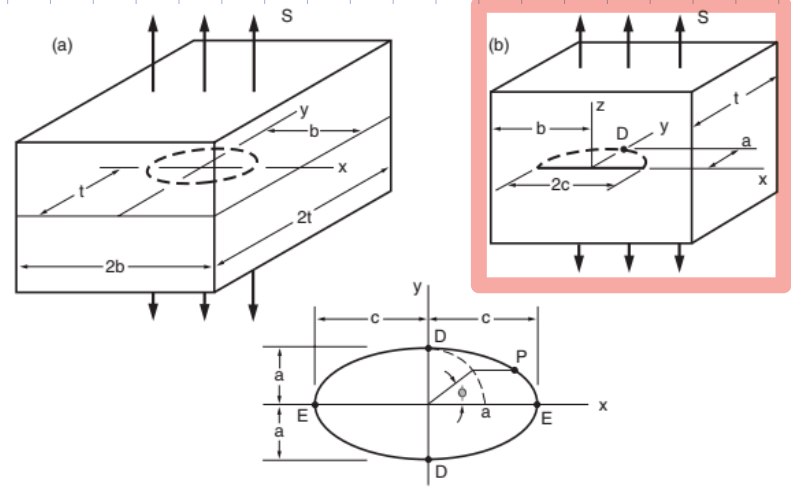
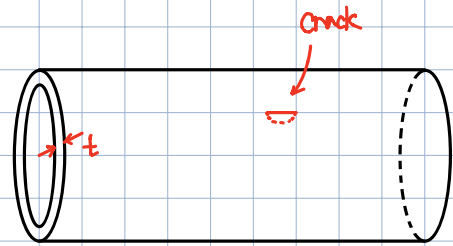
Double check! $\alpha = \sigma_b = \frac{18.3 \text{ mm}}{50 \text{ mm}} = 0.37 < 0.4 \rightarrow F = 1$ okay!

Example >

A pressure vessel (ASTM A517-F steel)
 RT $\hookrightarrow K_{Ic} = 187 \text{ MPa}\sqrt{\text{m}}$
 $\sigma_0 = 760 \text{ MPa}$

t = 50 mm
 A surface crack: A semi-elliptical shape
 surface length = $2c = 40 \text{ mm}$
 depth = $a = 10 \text{ mm}$

$S_z = 300 \text{ MPa}$ (normal to the crack plane)
 $S_x = 150 \text{ MPa}$ (parallel to the crack plane)



$$K_D = F_D S \sqrt{\frac{\pi a}{Q}}$$

$$Q \approx 1 + 1.464 \left(\frac{a}{c}\right)^{1.65} \quad (a/c \leq 1)$$

Case	Values for small $a/t, c/b$	Limits for 10% accuracy
(a)	$F_D = 1$	$a/t < 0.4, c/b < 0.2$
(b)	$F_D \approx 1.12$	$a/t < 0.3, c/b < 0.2$

Note: ¹Except limit to $a/t < 0.16$ if $a/c < 0.25$.

Figure 8.19 Stress intensity factors for (a) an embedded elliptical crack and (b) a similar half-elliptical surface crack. The equations give K_D at point D for a uniform tension normal to the crack plane. (Based on [Newman 86].)

Safety Factor ?

$$SF = \frac{K_{Ic}}{K} = \frac{187 \text{ MPa}\sqrt{\text{m}}}{49.2 \text{ MPa}\sqrt{\text{m}}} = 3.80 \text{ Ans.}$$

$$K = K_D = F_D S \sqrt{\frac{\pi a}{Q}}$$

$\leftarrow a/c = 10 \text{ mm} / 20 \text{ mm} = 0.5 < 1$

$$Q = 1 + (1.464) (a/c)^{1.65} = 1.466$$

$a/t = 10 \text{ mm} / 50 \text{ mm} = 0.2 < 0.3 \quad \therefore F_D \approx 1.12$

$$\therefore K = (1.12) \cdot (300 \text{ MPa}) \cdot \sqrt{\frac{(\pi) \cdot (0.010 \text{ m})}{1.466}} = 49.2 \text{ MPa}\sqrt{\text{m}}$$

Example >

A spherical pressure vessel (ASTM A517-F steel)

$$d_{in} = 1.5 \text{ m}$$

$$t = 10 \text{ mm}$$

$$p = 6 \text{ MPa}$$

$$K_{Ic} = 187 \text{ MPa}\sqrt{\text{m}}$$

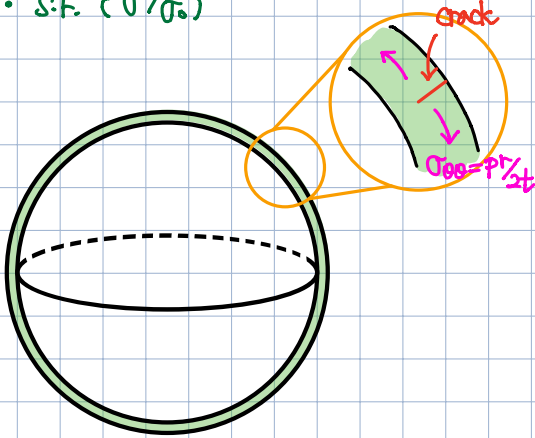
$$\sigma_0 = 760 \text{ MPa}$$

- the leak-before-break condition?
- S.F. (K/K_{Ic})
- S.F. (σ/σ_0)

Leak-Before-Break Design

- a thin-walled pressure vessel with a crack growing

1. The crack will gradually extend and penetrate wall
2. Sudden brittle fracture (before leaking)
↳ explosive release



$$\sigma_t = \frac{pr}{2t} = \frac{(6 \text{ MPa}) \cdot (750 \text{ mm})}{(2) \cdot (10 \text{ mm})} = 225 \text{ MPa}$$

$$K = FS\sqrt{\pi a} \quad \leftarrow F=1$$

↓

↳ the leak-before-break condition is met.

$$C_c = \frac{1}{\pi} \cdot \left(\frac{K_{Ic}}{\sigma_t} \right)^2 = \frac{1}{\pi} \cdot \left(\frac{187 \text{ MPa}\sqrt{\text{m}}}{225 \text{ MPa}} \right)^2 = 0.220 \text{ m} = \underline{220 \text{ mm}} \quad \text{Ans.}$$

if the vessel leaks,

$$K = FS\sqrt{\pi a} = (1) \cdot (225 \text{ MPa}) \cdot \sqrt{\pi \cdot (0.01 \text{ m})} = \underline{39.9 \text{ MPa}\sqrt{\text{m}}}$$

$$\therefore SF = \frac{K_{Ic}}{K} = \frac{187 \text{ MPa}\sqrt{\text{m}}}{39.9 \text{ MPa}\sqrt{\text{m}}} = \underline{4.69} \quad \text{Ans.}$$

$$SF = \frac{\sigma_0}{\sigma_t} = \frac{760 \text{ MPa}}{225 \text{ MPa}} = \underline{3.38} \quad \text{Ans.}$$