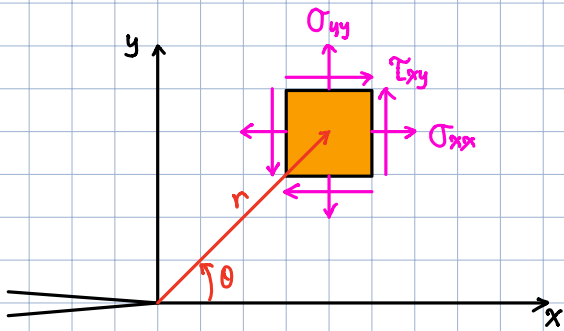


Defect-tolerant philosophy

The structural components are inherently flawed and the fatigue life is based on propagation of an initial flaw to a critical size

LEFM → The stress intensity factor, K

↳ the parameter to describe the stress/strains/displacements ahead of the crack tip.



Mode I

$$\sigma_{xx} = \frac{K_I}{\sqrt{2\pi r}} \cos \theta/2 \cdot (1 - \sin \theta/2 \cdot \sin 3\theta/2)$$

$$\sigma_{yy} = \frac{K_I}{\sqrt{2\pi r}} \cos \theta/2 \cdot (1 + \sin \theta/2 \cdot \sin 3\theta/2)$$

$$\tau_{xy} = \frac{K_I}{\sqrt{2\pi r}} \cos \theta/2 \cdot (\sin \theta/2 \cdot \sin 3\theta/2) + \dots$$

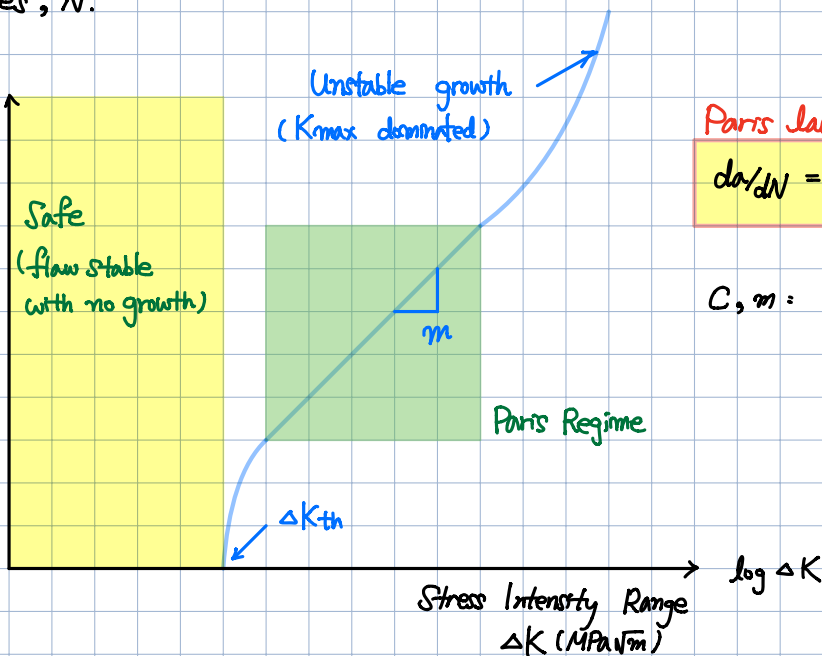
• **Concept:** The velocity of a moving fatigue crack subjected to constant stress amplitude loading is determined from the change in crack length, a , as a function of the number of loading cycles, N .

log da/dN
Fatigue-Crack Growth Rate
 da/dN (mm/cycle)

a = crack length

N = the number of loading cycles

da/dN : (The velocity)
The fatigue crack growth per loading cycle.



Paris law:

$$da/dN = C \cdot \Delta K^m$$

C, m = material constants

ΔK : the stress intensity factor range
 $\Delta K = K_{max} - K_{min}$

$K = f(\omega) \sigma^\infty \sqrt{\pi a}$ ←

1. $f(\omega)$ = Geometry
2. σ^∞ = far-field stress
3. a = crack length

Parrs law:

$$da/dN = C \cdot \Delta K^m$$

$$K = f(\sigma/w) \cdot \sigma \cdot \sqrt{\pi a}$$
$$\Delta K = f(\sigma/w) \cdot \Delta \sigma \cdot \sqrt{\pi a}$$

$$da/dN = C \cdot f(\sigma/w)^m \cdot \Delta \sigma^m \cdot \pi^{m/2} \cdot a^{m/2}$$

$$\int_{a_i}^{a_c} a^{-m/2} da = \int_0^{N_f} C \cdot f(\sigma/w)^m \cdot \Delta \sigma^m \cdot \pi^{m/2} \cdot dN$$

$$C \cdot f(\sigma/w)^m \cdot \Delta \sigma^m \cdot \pi^{m/2} \cdot N_f$$

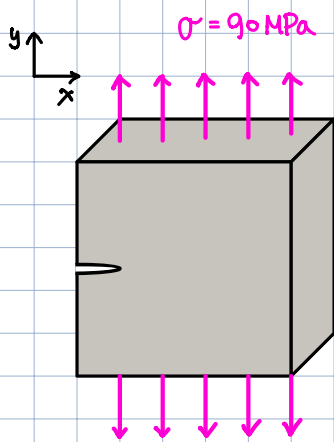
$$a^{-m/2+1} \cdot \frac{1}{-m/2+1} \Big]_{a_i}^{a_c}$$

$$= \frac{2}{2-m} \cdot \left(a_c^{\frac{2-m}{2}} - a_i^{\frac{2-m}{2}} \right)$$

$$= \frac{2}{m-2} \cdot \left(\frac{1}{a_i^{\frac{m-2}{2}}} - \frac{1}{a_c^{\frac{m-2}{2}}} \right)$$

$$\therefore N_f = \frac{2}{(m-2) C f(\sigma/w)^m \cdot \Delta \sigma^m \cdot \pi^{m/2}} \left(\frac{1}{a_i^{\frac{m-2}{2}}} - \frac{1}{a_c^{\frac{m-2}{2}}} \right) \quad \text{for } m \neq 2$$

Example >



a fracture toughness : $9.5 \text{ MPa}\sqrt{\text{m}}$
 $C = 6 \times 10^{-11} (\text{MPa}\sqrt{\text{m}})^m$
 $m = 4$

$$K_I = (1.12) \cdot \sigma \sqrt{\pi a}$$

initial crack length : 1 mm

10^5 cycles per month

a) What is the critical flaw size ?

b) How many fatigue cycles will this system last ?

a) $\Delta K = (1.12) \cdot \Delta \sigma \cdot \sqrt{\pi a}$

$$a_{cr} = \frac{1}{\pi} \cdot \frac{\Delta K^2}{(1.12)^2 \Delta \sigma^2} = \frac{1}{\pi} \cdot \left\{ \frac{9.5 \text{ MPa}\sqrt{\text{m}}}{(1.12) \cdot (90 \text{ MPa})} \right\}^2$$
$$= 0.028 \text{ m} = \underline{28 \text{ mm}}$$

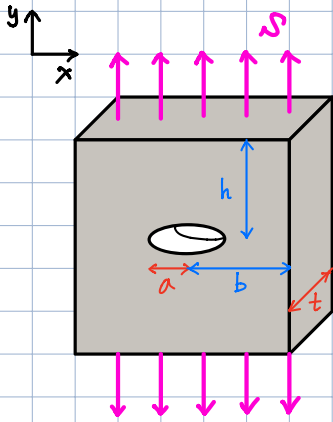
$$\therefore N_f = \frac{2}{(m-2) C f(a/w)^m \cdot \Delta \sigma^m \cdot \pi^{m/2}} \left(\frac{1}{a_i^{m/2}} - \frac{1}{a_c^{m/2}} \right) \quad \text{for } m \neq 2$$

$$= \frac{2}{(4-2) \cdot (6 \times 10^{-11}) \cdot (1.12)^4 \cdot (90)^4 \cdot \pi^{4/2}} \cdot \left(\frac{1}{(0.001)^1} - \frac{1}{(0.028)^1} \right)$$

$$= 1.57 \times 10^4 \text{ cycles} \approx \underline{15.7 \text{ months}} \text{ Ans.}$$

Example >

a center-cracked plate : $b = 38 \text{ mm}$
 $t = 6 \text{ mm}$
 $a_i = 1 \text{ mm}$



cyclic tensile loading : $P_0 - 240 \text{ kN}$ (freq = 5 Hz)

a) what is the critical crack length, $a_c = ?$ ($K_{Ic} = 130 \text{ MPa}\sqrt{\text{m}}$)

$$\sigma = \frac{P}{2bt} \rightarrow \sigma_{\max} = \frac{240 \text{ kN}}{(2)(0.038 \text{ m}) \cdot (0.006 \text{ m})} = 526.3 \text{ MPa}$$

$$K = f(\alpha/w) \sigma \sqrt{\pi a} \rightarrow K_{Ic} = f(\alpha/w) \sigma \sqrt{\pi a_c} \rightarrow a_c = \frac{1}{\pi} \cdot \left\{ \frac{K_{Ic}}{f(\alpha/w) \cdot \sigma} \right\}^2 \leftarrow \text{a center cracked plate}$$

$f(\alpha/w) = 1$

$$\therefore a_c = \frac{1}{\pi} \cdot \left\{ \frac{130 \text{ MPa}\sqrt{\text{m}}}{(1) \cdot (526.3 \text{ MPa})} \right\}^2 = \underline{19.4 \text{ mm}} \text{ Ans.}$$

b) How many cycles can be applied before failure occurs if life is estimated from its Paris law behaviors?

$$da/dN = C(\Delta K)^m, \quad C = 1.095 \times 10^{-12} \text{ (da/dN}/\Delta K^m), \quad m = 3.24$$

$$\Delta K = f(\alpha/w) \Delta \sigma \sqrt{\pi a} \leftarrow \Delta \sigma = \sigma_{\max} - \sigma_{\min} \leftarrow \begin{aligned} \sigma_{\max} &= 526.3 \text{ MPa} \\ \sigma_{\min} &= \frac{80 \text{ kN}}{(2)(0.038 \text{ m}) \cdot (0.006 \text{ m})} = 175.4 \text{ MPa} \end{aligned}$$

$$\therefore \Delta K = (1) \cdot (526.3 - 175.4) \cdot \sqrt{\pi a}$$

$$= 357 \text{ MPa}\sqrt{\pi a}$$

Paris law:

$$da/dN = C \cdot \Delta K^m = (1.095 \times 10^{-12} \frac{\text{m/cycle}}{(\text{MPa}\sqrt{\text{m}})^{3.24}}) \cdot (357 \text{ MPa}\sqrt{\pi a})^{3.24}$$

$$= \text{⊙} a^{1.62}$$

$$\therefore \int_{a_i}^{a_c} a^{-1.62} da = \int_0^{N_f} \text{⊙} dN$$

$$\int_{1 \times 10^{-3}}^{19.4 \times 10^{-3}} a^{-1.62} da = \int_0^{N_f} 0.0013 dN = (0.0013) \cdot N_f$$

$$a^{-0.62} \cdot \frac{1}{-0.62} \Big|_{1 \times 10^{-3}}^{19.4 \times 10^{-3}} = (0.0013) N_f$$

$$98.26 = (0.0013) N_f$$

$$\therefore N_f = \underline{75,562 \text{ cycles}} \text{ Ans.}$$